

FINAL TEST - ANALYTICAL MECHANICS - OCTOBER, 19th 2016

SOLUTIONS

Problem 1.

$$(1) \quad \det \mathbf{F} = (\mathbf{F}\mathbf{e}_1 \times \mathbf{F}\mathbf{e}_2) \cdot \mathbf{F}\mathbf{e}_3 = (\mathbf{e}_1 \times (-2\mathbf{e}_1 + \mathbf{e}_2)) \cdot (3\mathbf{e}_2 + \mathbf{e}_3) = 1$$
$$\operatorname{tr}\mathbf{F} = 3$$

$$(2) \quad \mathbf{F}\mathbf{u} = -2\mathbf{e}_1 + 3\mathbf{e}_2 - 6\mathbf{e}_1 + \mathbf{e}_3 + 3\mathbf{e}_2 = -8\mathbf{e}_1 + 6\mathbf{e}_2 + \mathbf{e}_3$$

$$(3) \quad \operatorname{sym}\mathbf{F} = \mathbf{I} - \mathbf{e}_1 \otimes \mathbf{e}_2 - \mathbf{e}_2 \otimes \mathbf{e}_1 + \frac{3}{2}\mathbf{e}_2 \otimes \mathbf{e}_3 + \frac{3}{2}\mathbf{e}_3 \otimes \mathbf{e}_2$$
$$\operatorname{skw}\mathbf{F} = -\mathbf{e}_1 \otimes \mathbf{e}_2 + \mathbf{e}_2 \otimes \mathbf{e}_1 + \frac{3}{2}\mathbf{e}_2 \otimes \mathbf{e}_3 - \frac{3}{2}\mathbf{e}_3 \otimes \mathbf{e}_2$$

$$(4) \quad \mathbf{w}(\mathbf{F}) = -\frac{3}{2}\mathbf{e}_1 + \mathbf{e}_3$$

$$(5) \quad \mathbf{F}^{-1} = \mathbf{I} + 2\mathbf{e}_1 \otimes \mathbf{e}_2 - 6\mathbf{e}_1 \otimes \mathbf{e}_3 - 3\mathbf{e}_2 \otimes \mathbf{e}_3$$
$$\mathbf{F}^* = (\det \mathbf{F})\mathbf{F}^{-\top} = \mathbf{I} + 2\mathbf{e}_2 \otimes \mathbf{e}_1 - 6\mathbf{e}_3 \otimes \mathbf{e}_1 - 3\mathbf{e}_3 \otimes \mathbf{e}_2$$

$$(6) \quad |\mathbf{F}^*\mathbf{e}_1| = |\mathbf{e}_1 + 2\mathbf{e}_2 - 6\mathbf{e}_3| = \sqrt{41}$$

Problem 2.

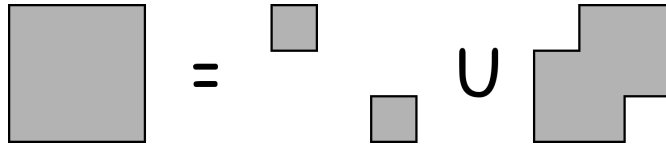
(1) By symmetry,

$$C - O = \frac{3}{2}L(\mathbf{e}_x + \mathbf{e}_y)$$

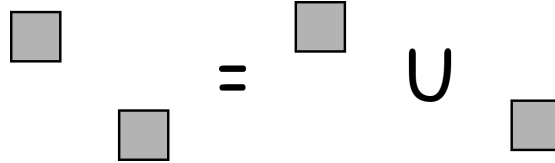
(2) and a principal basis of \mathbf{I}_C is

$$\left(\mathbf{e}_1 := \frac{\sqrt{2}}{2}(\mathbf{e}_x + \mathbf{e}_y), \frac{\sqrt{2}}{2}(-\mathbf{e}_x + \mathbf{e}_y), \mathbf{e}_z := \mathbf{e}_x \times \mathbf{e}_y \right).$$

(3) To compute \mathbf{I}_C we can use the Composition Theorem, indeed:



Therefore $\mathbf{I}_C = \mathbf{I}_{\text{BigSquare}C} - \mathbf{I}_{\text{TwoSquares}C}$, since the center of mass of both BigSquare and TwoSquares is C . Moreover



and hence, by the Composition Theorem again,

$$\mathbf{I}_{\text{TwoSquares}C} = \mathbf{I}_{\text{Square}1C_1} + \mathbf{I}_{\text{Square}2C_2} + \frac{m_1 m_2}{m_1 + m_2} |C_1 - C_2|^2 (\mathbf{I} - \mathbf{e}_2 \otimes \mathbf{e}_2)$$

where $|C_1 - C_2|^2 = 8L^2$, $m_1 = m_2 = \rho L^2$ and $m = 7\rho L^2$.

We now need to compute the central tensor of inertia for a generic square with mass density ρ and edge a . We call A the center of mass of the generic square, then by symmetry

$$I_{Axx} = I_{Ayy} = 4\rho \int_0^{\frac{a}{2}} \int_0^{\frac{a}{2}} x^2 dx dy = 2a\rho \int_0^{\frac{a}{2}} x^2 dx = \frac{1}{12}\rho a^4 = \frac{1}{2}I_{Azz}.$$

Since $I_{Axx} = I_{Ayy}$, all the directions on the plane span($\mathbf{e}_x, \mathbf{e}_y$) are principal, and then

$$I_{A11} = I_{A22} = 4\rho \int_0^{\frac{a}{2}} \int_0^{\frac{a}{2}} x^2 dx dy = 2a\rho \int_0^{\frac{a}{2}} x^2 dx = \frac{1}{12}\rho a^4 = \frac{1}{2}I_{Azz}.$$

Therefore

$$\mathbf{I}_{\text{Square}1C_1} = \frac{1}{12}\rho L^4 (\mathbf{e}_1 \otimes \mathbf{e}_1 + \mathbf{e}_2 \otimes \mathbf{e}_2 + 2\mathbf{e}_z \otimes \mathbf{e}_z)$$

$$\mathbf{I}_{\text{Square}2C_2} = \frac{1}{12}\rho L^4 (\mathbf{e}_1 \otimes \mathbf{e}_1 + \mathbf{e}_2 \otimes \mathbf{e}_2 + 2\mathbf{e}_z \otimes \mathbf{e}_z)$$

whence

$$\begin{aligned}\mathbf{I}_{\text{TwoSquares}C} &= \frac{1}{6}\rho L^4(\mathbf{e}_1 \otimes \mathbf{e}_1 + \mathbf{e}_2 \otimes \mathbf{e}_2 + 2\mathbf{e}_z \otimes \mathbf{e}_z) + 4\rho L^4(\mathbf{I} - \mathbf{e}_2 \otimes \mathbf{e}_2) = \\ &= \frac{25}{6}\rho L^4\mathbf{e}_1 \otimes \mathbf{e}_1 + \frac{1}{6}\rho L^4\mathbf{e}_2 \otimes \mathbf{e}_2 + \frac{13}{3}\rho L^4\mathbf{e}_z \otimes \mathbf{e}_z\end{aligned}$$

and

$$\begin{aligned}\mathbf{I}_C &= \frac{81}{12}\rho L^4(\mathbf{e}_1 \otimes \mathbf{e}_1 + \mathbf{e}_2 \otimes \mathbf{e}_2 + 2\mathbf{e}_z \otimes \mathbf{e}_z) - \frac{25}{6}\rho L^4\mathbf{e}_1 \otimes \mathbf{e}_1 - \frac{1}{6}\rho L^4\mathbf{e}_2 \otimes \mathbf{e}_2 - \frac{13}{3}\rho L^4\mathbf{e}_z \otimes \mathbf{e}_z = \\ &= \frac{31}{12}\rho L^4\mathbf{e}_1 \otimes \mathbf{e}_1 + \frac{79}{12}\rho L^4\mathbf{e}_2 \otimes \mathbf{e}_2 + \frac{55}{6}\rho L^4\mathbf{e}_z \otimes \mathbf{e}_z =\end{aligned}$$

(4) By Huygens-Steiner's Theorem

$$\begin{aligned}\mathbf{I}_O &= \mathbf{I}_C + m|C - O|^2(\mathbf{I} - \mathbf{e}_1 \otimes \mathbf{e}_1) = \\ &= \frac{31}{12}\rho L^4\mathbf{e}_1 \otimes \mathbf{e}_1 + \frac{79}{12}\rho L^4\mathbf{e}_2 \otimes \mathbf{e}_2 + \frac{55}{6}\rho L^4\mathbf{e}_z \otimes \mathbf{e}_z + \frac{63}{2}\rho L^4(\mathbf{I} - \mathbf{e}_1 \otimes \mathbf{e}_1) = \\ &= \frac{31}{12}\rho L^4\mathbf{e}_1 \otimes \mathbf{e}_1 + \frac{457}{12}\rho L^4\mathbf{e}_2 \otimes \mathbf{e}_2 + \frac{244}{6}\rho L^4\mathbf{e}_z \otimes \mathbf{e}_z\end{aligned}$$

Problem 3.

(1) The arc-length parameter is

$$\begin{aligned}
 s(t) &= \int |\dot{P}(u)| du = \\
 &= \int |(-\sinh(t) \sin(\cosh(t)), \sinh(t) \cos(\cosh(t)), 1)| du = \\
 &= \int \sqrt{\sinh^2(t) + 1} du = \\
 &= \int \sqrt{\cosh^2(t)} du = \\
 &= \sinh(t)
 \end{aligned}$$

(2) Therefore $t(s) = \operatorname{asinh}(s)$ and the arc-length parametrization of the curve is

$$Q(s) = (\cos(\cosh(\operatorname{asinh}(s))), \sin(\cosh(\operatorname{asinh}(s))), \operatorname{asinh}(s))$$

(3) Then

$$\mathbf{t}(s) = Q'(s) = \left(\frac{-s \sin(\cosh(\operatorname{asinh}(s)))}{\sqrt{s^2 + 1}}, \frac{s \cos(\cosh(\operatorname{asinh}(s)))}{\sqrt{s^2 + 1}}, \frac{1}{\sqrt{s^2 + 1}} \right).$$

Since $\frac{d}{ds} \frac{1}{\sqrt{s^2 + 1}} = -\frac{s}{(\sqrt{s^2 + 1})^3}$ and $\frac{d}{ds} \frac{s}{\sqrt{s^2 + 1}} = \frac{1}{\sqrt{s^2 + 1}} - \frac{s^2}{(\sqrt{s^2 + 1})^3} = \frac{1}{(\sqrt{s^2 + 1})^3}$ we get

$$\begin{aligned}
 \frac{d}{ds} \frac{-s \sin(\cosh(\operatorname{asinh}(s)))}{\sqrt{s^2 + 1}} &= -\frac{\sin(\cosh(\operatorname{asinh}(s)))}{(\sqrt{s^2 + 1})^3} - \frac{s^2 \cos(\cosh(\operatorname{asinh}(s)))}{(\sqrt{s^2 + 1})^2} \\
 \frac{d}{ds} \frac{s \cos(\cosh(\operatorname{asinh}(s)))}{\sqrt{s^2 + 1}} &= \frac{\cos(\cosh(\operatorname{asinh}(s)))}{(\sqrt{s^2 + 1})^3} - \frac{s^2 \sin(\cosh(\operatorname{asinh}(s)))}{(\sqrt{s^2 + 1})^2} \\
 \frac{d}{ds} \frac{1}{\sqrt{s^2 + 1}} &= -\frac{s}{(\sqrt{s^2 + 1})^3}
 \end{aligned}$$

whence

$$c(s) = |\mathbf{t}'(s)| = |Q''(s)| = \sqrt{\frac{1}{(s^2 + 1)^3} + \frac{s^4}{(s^2 + 1)^2} + \frac{s^2}{(s^2 + 1)^3}} = \frac{\sqrt{s^4 + 1}}{s^2 + 1}.$$

Problem 4.

- (1) Let $P(x) = (x, y(x))$ be the curve that describes the shape of the cable at equilibrium.

Then the tangent vector to the curve is $\mathbf{t}(x) = \frac{1}{\sqrt{1 + \dot{y}^2(x)}}(1, \dot{y}(x))$.

Since the force \mathbf{F}_O is parallel to OB and the stress $\boldsymbol{\varphi}(0) = -\mathbf{F}_O$ at O is equal to $T(0)\mathbf{t}(0)$ (where $T > 0$ is the tension), the tangent $\mathbf{t}(0)$ is parallel to \mathbf{F}_O too (and with opposite direction).

Therefore, $\mathbf{t}(0) = (\cos \alpha, -\sin \alpha)$

and $\mathbf{t}(0) = \frac{1}{\sqrt{1 + \dot{y}^2(0)}}(1, \dot{y}(0))$

by equating the two components and dividing, we obtain

$$\dot{y}(0) = \frac{-\sin \alpha}{\cos \alpha} = -\tan \alpha.$$

- (2) Note that $y(0) = y(x_0) = 0$. Since the only distributed force is the gravitational force due to the weight of the cable, the shape at equilibrium is

$$y(x) = \frac{K}{\lambda g} \cosh \left(\frac{\lambda g}{K} x - C_1 \right) - C_2.$$

From the condition $y(0) = 0$ we get $C_2 = \frac{K}{\lambda g} \cosh(C_1)$. Given that $\dot{y}(x) = \sinh \left(\frac{\lambda g}{K} x - C_1 \right)$, from $\dot{y}(0) = -\tan \alpha$ we get $-\tan \alpha = \sinh(-C_1)$, whence $C_1 = \operatorname{asinh}(\tan \alpha)$ and

$$y(x) = \frac{K}{\lambda g} \cosh \left(\frac{\lambda g}{K} x - \operatorname{asinh}(\tan \alpha) \right) - \frac{K}{\lambda g} \cosh(\operatorname{asinh}(\tan \alpha)).$$

From the condition $y(x_0) = 0$ we get

$$\frac{K}{\lambda g} \cosh \left(\frac{\lambda g}{K} x_0 - \operatorname{asinh}(\tan \alpha) \right) = \frac{K}{\lambda g} \cosh(\operatorname{asinh}(\tan \alpha))$$

which means that $\frac{\lambda g}{K} x_0 - \operatorname{asinh}(\tan \alpha)$ and $\operatorname{asinh}(\tan \alpha)$ have the same hyperbolic cosine.

Therefore: either $\frac{\lambda g}{K} x_0 - \operatorname{asinh}(\tan \alpha) = \operatorname{asinh}(\tan \alpha)$ and $x_0 = \frac{2K}{\lambda g} \operatorname{asinh}(\tan \alpha)$, or

$\frac{\lambda g}{K} x_0 - \operatorname{asinh}(\tan \alpha) = -\operatorname{asinh}(\tan \alpha)$ that yields to $x_0 = 0$ (which does not make any sense in this problem). In conclusion,

$$x_0 = \frac{2K}{\lambda g} \operatorname{asinh}(\tan \alpha)$$

- (3) The known length of the cable entails

$$\begin{aligned} L &= \int_0^{x_0} \sqrt{1 + \dot{y}^2(x)} dx = \int_0^{x_0} \cosh \left(\frac{\lambda g}{K} x - \frac{\lambda g}{2K} x_0 \right) dx = \\ &= \frac{K}{\lambda g} \sinh \left(\frac{\lambda g}{K} x_0 - \frac{\lambda g}{2K} x_0 \right) - \frac{K}{\lambda g} \sinh \left(-\frac{\lambda g}{2K} x_0 \right) = \\ &= 2 \frac{K}{\lambda g} \sinh \left(\frac{\lambda g}{2K} x_0 \right) = 2 \frac{K}{\lambda g} \tan \alpha \end{aligned}$$

whence

$$K = \frac{\lambda g L}{2 \tan \alpha}$$

(4) and

$$y(x) = \frac{L}{2 \tan \alpha} \cosh \left(\frac{2 \tan \alpha}{L} x - \operatorname{asinh}(\tan \alpha) \right) - \frac{L}{2 \tan \alpha} \cosh (\operatorname{asinh}(\tan \alpha)).$$

(5) Since $x_0 = \frac{L}{\tan \alpha} \operatorname{asinh}(\tan \alpha) = 2l \cos \alpha$ then

$$l = \frac{L \operatorname{asinh}(\tan \alpha)}{2 \sin \alpha}.$$

(6) Since $T(0) = K \sqrt{1 + \dot{y}^2(0)}$, we obtain

$$|\mathbf{F}_O| = T(0) = \frac{\lambda g L}{2 \tan \alpha} \sqrt{1 + \tan^2 \alpha} = \frac{\lambda g L}{2 \sin \alpha}$$

Problem 5.

For any $s \in [0, 2L]$, let $\vartheta(s)$ be the angle between \mathbf{e}_x and the tangent vector $t(s)$.

SETUP:

$$\begin{cases} \mathbf{d}_2(s) = \mathbf{e}_z \\ \mathbf{d}_3(s) = \mathbf{t}(s) = \cos \vartheta(s) \mathbf{e}_x + \sin \vartheta(s) \mathbf{e}_y \\ \mathbf{d}_1(s) = \mathbf{d}_2(s) \times \mathbf{d}_3(s) = -\sin \vartheta(s) \mathbf{e}_x + \cos \vartheta(s) \mathbf{e}_y \end{cases} \quad \text{then} \quad \begin{cases} \mathbf{d}'_2(s) = \mathbf{0} \\ \mathbf{d}'_3(s) = \vartheta'(s) \mathbf{d}_1(s) \end{cases}$$

From $\mathbf{d}'_i(s) = \mathbf{u}(s) \times \mathbf{d}_i$, we obtain $\mathbf{u} = \vartheta'(s) \mathbf{d}_2(s)$ and hence $\gamma(s) = B\vartheta'(s) \mathbf{e}_z$.

(1) BOUNDARY CONDITIONS:

$$O : \begin{cases} y(0) = 0 \\ \varphi(0) = -\mathbf{F}_O \text{ [to be determined]} \\ \gamma(0) = -\mathbf{G} = -2FL\mathbf{e}_z \Rightarrow \vartheta'(0) = -\frac{2FL}{B} \end{cases} \quad 2L : \begin{cases} y(2L) = 0 \\ \varphi(2L) = \mathbf{F}_{2L} = \phi \mathbf{e}_y \text{ [\phi to be determined]} \\ \gamma(2L) = \mathbf{0} \Rightarrow \vartheta'(2L) = 0 \end{cases}$$

(2) CONDITIONS AT $s = L$:

$$s = L : \begin{cases} \llbracket \varphi(L) \rrbracket = -\mathbf{F} = 4F\mathbf{e}_y \\ \llbracket \gamma(L) \rrbracket = \mathbf{0} \Rightarrow \llbracket \vartheta'(L) \rrbracket = 0 \Rightarrow \vartheta' \text{ continuous at } s = L \text{ (hence everywhere)} \end{cases}$$

(3) EQUILIBRIUM EQUATIONS FOR FORCES ACTING ON THE BEAM:

$$\mathbf{F} + \mathbf{F}_O + \mathbf{F}_{2L} = \mathbf{0} \quad \Rightarrow \quad -4F\mathbf{e}_y + \mathbf{F}_O + \phi \mathbf{e}_y = \mathbf{0}$$

hence

$$\mathbf{F}_O = (4F - \phi) \mathbf{e}_y$$

EQUILIBRIUM EQUATIONS FOR TORQUES ACTING ON THE BEAM W.R.T. O:

$$\mathbf{G} + L\mathbf{e}_x \times \mathbf{F} + 2L\mathbf{e}_x \times \mathbf{F}_{2L} = \mathbf{0} \quad \Rightarrow \quad 2FL\mathbf{e}_z - 4FL\mathbf{e}_z + 2\phi L\mathbf{e}_y = \mathbf{0}$$

hence $\phi = F$ and the REACTIVE FORCES are

$$\begin{cases} \mathbf{F}_{2L} = F\mathbf{e}_y \\ \mathbf{F}_O = 3F\mathbf{e}_y \end{cases}$$

(4) EQUILIBRIUM EQUATIONS FOR CONTACT FORCES:

$$\begin{cases} \varphi'(s) = \mathbf{0} & s \in [0, L) \\ \varphi'(s) = \mathbf{0} & s \in (L, 2L] \end{cases} \quad \text{then} \quad \begin{cases} \varphi(s) = \text{constant} = \varphi(0) = -\mathbf{F}_O = -3F\mathbf{e}_y & s \in [0, L) \\ \varphi(s) = \text{constant} = \varphi(2L) = \mathbf{F}_{2L} = F\mathbf{e}_y & s \in (L, 2L] \end{cases}$$

EQUILIBRIUM EQUATIONS FOR CONTACT TORQUES:

$$\begin{cases} \gamma'(s) - 3F \cos \vartheta(s) \mathbf{e}_z = \mathbf{0} & s \in [0, L) \\ \llbracket \gamma(L) \rrbracket = \mathbf{0} \\ \gamma'(s) + F \cos \vartheta(s) \mathbf{e}_z = \mathbf{0} & s \in (L, 2L] \end{cases} \quad \text{whence} \quad \begin{cases} B\vartheta''(s) - 3F \cos \vartheta(s) = 0 & s \in [0, L) \\ \llbracket \vartheta'(L) \rrbracket = 0 \\ B\vartheta''(s) + F \cos \vartheta(s) = 0 & s \in (L, 2L] \end{cases}$$

SMALL DEFLECTIONS APPROXIMATION: We suppose $|\vartheta(s)| \ll 1$ for all $s \in [0, 2L]$.

Then we can approximate $\sin \vartheta(s)$ with $\vartheta(s)$ and $\cos \vartheta(s)$ with 1.

Then $s \approx x$ and $y'(x) \approx \vartheta(x)$.

Hence

$$\begin{cases} B\vartheta''(x) - 3F = 0 & x \in [0, L] \\ \vartheta' \text{ continuous at } x = L \\ B\vartheta''(x) + F = 0 & x \in (L, 2L] \end{cases}$$

We start integrating and using the boundary conditions:

$$\begin{cases} B\vartheta'(x) = 3Fx + C_1 & x \in [0, L] \\ B\vartheta'(x) = -Fx + C_2 & x \in (L, 2L] \end{cases} \quad \text{and} \quad \begin{cases} \vartheta'(0) = -\frac{2FL}{B} \\ \vartheta'(2L) = 0 \end{cases}$$

whence $\begin{cases} B\vartheta'(x) = 3Fx - 2FL & x \in [0, L] \\ B\vartheta'(x) = -Fx + 2FL & x \in [L, 2L] \end{cases}$ and $\begin{cases} B\vartheta(x) = \frac{3F}{2}x^2 - 2FLx + C_3 & x \in [0, L] \\ B\vartheta(x) = -\frac{F}{2}x^2 + 2FLx + C_4 & x \in [L, 2L] \end{cases}$

By the continuity of $\vartheta(x)$ (at $x = L$), we get $C_4 = -2FL^2 + C_3$, then

$$\begin{cases} B\vartheta(x) = \frac{3F}{2}x^2 - 2FLx + C_3 & x \in [0, L] \\ B\vartheta(x) = -\frac{F}{2}x^2 + 2FLx - 2FL^2 + C_3 & x \in [L, 2L] \end{cases}$$

and $\begin{cases} By(x) = \frac{F}{2}x^3 - FLx^2 + C_3x + C_5 & x \in [0, L] \\ By(x) = -\frac{F}{6}x^3 + FLx^2 - 2FL^2x + C_3x + C_6 & x \in [L, 2L] \end{cases}$

From the boundary conditions $y(0) = 0$ and $y(2L) = 0$ we get

$$\text{and} \quad \begin{cases} By(x) = \frac{F}{2}x^3 - FLx^2 + C_3x & x \in [0, L] \\ By(x) = -\frac{F}{6}x^3 + FLx^2 - 2FL^2x + C_3x - 2C_3L + \frac{4F}{3}L^3 & x \in [L, 2L] \end{cases}$$

Equating the right sides of the last two equations for $x = L$ (because $y(x)$ is continuous), we obtain $C_3 = \frac{1}{3}FL^2$ and hence the function $y(x)$ which represents the shape of the center line of the beam at equilibrium in case of small deflections is

$$\begin{cases} y(x) = \frac{F}{2B}x \left(x^2 - 2Lx + \frac{2}{3}L^2 \right) & x \in [0, L] \\ y(x) = -\frac{F}{6B}(x^3 - 6Lx^2 + 10L^2x - 4L^3) = -\frac{F}{6B}(x - 2L)(x^2 - 4Lx + 2L^2) & x \in [L, 2L] \end{cases}$$

(5) The function $\vartheta(x) \approx y'(x)$ describes the slope of the center line of the curve at equilibrium

$$\begin{cases} \vartheta(x) = \frac{3F}{2B} \left(x^2 - \frac{4}{3}Lx + \frac{2}{9}L^2 \right) = \frac{3F}{2B} \left(x - \frac{2 + \sqrt{2}}{3}L \right) \left(x - \frac{2 - \sqrt{2}}{3}L \right) & x \in [0, L] \\ \vartheta(x) = -\frac{F}{2B} \left(x^2 - 4Lx + \frac{10}{3}L^2 \right) = -\frac{F}{2B} \left(x - 2L - \sqrt{\frac{2}{3}}L \right) \left(x - 2L + \sqrt{\frac{2}{3}}L \right) & x \in [L, 2L] \end{cases}$$

and

$$\begin{cases} \vartheta'(x) = \frac{F}{B}(3x - 2L) & x \in [0, L] \\ \vartheta'(x) = -\frac{F}{B}(x - 2L) & x \in [L, 2L] \end{cases}$$

Therefore $\vartheta(x)$ is decreasing from $\vartheta(0) = \frac{FL^2}{3B}$ to $\vartheta\left(\frac{2}{3}L\right) = -\frac{FL^2}{3B}$ and then increasing to $\vartheta(2L) = \frac{FL^2}{3B}$.

Then

$$\vartheta_{\max} = \frac{FL^2}{3B}.$$

(6) The approximation with small deflections is reliable for $\vartheta_{\max} \ll 1$, whence for

$$F \ll 3\frac{B}{L^2}.$$

NOTE: $y(x)$ is increasing from $y(0) = 0$ to $y\left(\frac{2 - \sqrt{2}}{3}L\right) = 2\frac{\sqrt{2} - 1}{27}\frac{FL^3}{B}$, then decreasing to $y\left(2L - \sqrt{\frac{2}{3}}L\right) = -\frac{2}{9}\sqrt{\frac{2}{3}}\frac{FL^3}{B}$, and then increasing to $y(2L) = 0$. Then the maximum deflection is

$$\delta_{\max} = \left| y\left(2L - \sqrt{\frac{2}{3}}L\right) \right| = \frac{2}{9}\sqrt{\frac{2}{3}}\frac{FL^3}{B}.$$